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Vibration analysis and measurement based on defect signal evaluation: Gas turbine investigation

Boulanouar Saadat^{1*}, Ahmed Hafaifa² and Mouloud Guemana³

¹ Applied Automation and Industrial Diagnostics Laboratory, Faculty of Science and Technology, University of Djelfa 17000 DZ, Algeria.

² Faculty of Science and Technology, University of Médéa, 26000, Algeria

Abstract. The vibration behavior of rotating machines is a very important industrial challenge, can cause aging and tiredness affect its components. In this work, we propose to study the gas turbine vibration behavior based on defect signal evaluation, an analysis made by the vibration signals measured. The tests results realized on the examined gas turbine shows that the vibration effects observed provide a very powerful tool for protected the mechanical condition of the examined gas turbine

Keywords: Vibration behavior, rotating machines vibration analysis, gear defect evaluation, gas turbine, measures validation

1. Introduction

Technological evolution of industrial equipment in recent years, allows to the vibration monitoring become the primary tool of conditional preventative maintenance [1, 2, 6, 17, 26 and 32]. There are many techniques that can be used in a maintenance program, such this techniques the vibration analysis [4, 8, 10, 15 and 21]. This vibration analysis technique allows detecting virtually all defects that may appear in rotating machines [16, 18, 23, 29 and 30]: Unbalance, alignment defects, worn or damaged bearing ... etc. This result in a variation of internal forces experienced by the machine, and thus to a change in its vibration behavior. The vibration analysis has become the main technique used in the management of the condition-based maintenance. The largest share of industrial equipment is mechanical in nature, this technique has many potential applications and provides the best benefits in a program intended to cover an entire factory [7, 14, 22 and 31]. It uses the noise or vibration created by mechanical equipment (and in some cases by industrial systems) to determine the actual operating mode [3, 5].

In this context, we propose in this work to examine and illustrate the ability of the application of the methods of vibration analysis based on gear defect evaluation to monitor the operating condition of a gas turbine. This is part of predictive maintenance policy tool in industrial production, taking the example of a gas turbine system type GE MS 3002. Indeed, the vibration analysis of rotating machinery is now widely used by manufacturers to diagnose faults on their machines before it will undergo a fortuitous is conditional maintenance. We show in this work, through various tests, the results clearly show the detection of malfunctions of rotating machinery in the examined gas turbine, and allowing for better performance during its operation for the maintenance strategy.

^{*} Corresponding author.

E-mail: N_sadat@univ-djelfa.dz (Saadat B.).

AAIDL, Faculty of Science and Technology, University of Djelfa, 17000, Algeria.

2. Vibration analysis

The exploitation of rotating machinery always produced effort (turning effort, turbulence, shocks, volatility) that will often cause subsequent failures [9, 12, 19 and 28]. To establish a diagnosis, it must rely on the fact that a machine emits outward many signals that are symptomatic of its operation, such as heat, power consumption, noise, vibration ... etc. Vibration analysis plays an important role not only in the diagnosis of the state machines for servicing, is the starting point for any diagnosis vibration [11, 20]. One can make the succeeding exchange suspicious parts: drive motor, rods, bearings and balancing all the rotating parts [13, 24 and 25]. Based on vibration analysis, we can make successive trading decisions suspicious parts: drive motor, coupling, bearings, and balance all rotating parts. It is efficient and economical separation with appropriate means according to different vibrations components, undertake targeted maintenance. The effective vibration amplitude is defined mathematically by the equation (1) and is expressed as a function of the peak amplitude.

$$\mathbf{A}_{\rm eff} = \sqrt{\frac{1}{T} \int_{0}^{T} a_{\rm eff}^{2}} dt$$
 (1)

Where , a(t) is the instantaneous amplitude of the vibration signal, T is the length of the vibration signal and A_c the peak amplitude.

The displacement x(t) of a harmonic vibration and written by the flowing equation:

$$x(t) = A\sin(\omega t + \varphi) \qquad \qquad \mathbf{A}_{\rm eff} = A_c \, \frac{\sqrt{2}}{2} \tag{2}$$

Where ω is the pulsation, φ is the phase and A is the amplitude.

The speed and acceleration of the vibration is obtained by the displacement differentiating:

$$X(t) = A\sin(2\pi f t) \Rightarrow V(t) = \frac{dX}{dt} \Rightarrow V(t) = 2\pi f A \sin(2\pi f t + \pi/2)$$
$$\gamma(t) = \frac{d^2 X}{dt^2} \Rightarrow \gamma(t) = (2\pi f)^2 A \sin(2\pi f t \pm \pi)$$

$$w = 2\pi f = \frac{2\pi}{T}$$
(3)

Where T is the period express (s) and f is the frequency in hertz (Hz)

The vibration signal delivered by a sensor may be represented in different ways [6, 27 and 32]; the first that comes to mind is the representation function of time. This representation is also used to monitor the vibration behavior of a machine, based on its operating parameters. For it to be interpreted, the signal must be decomposed into different elementary sinusoidal components. If this decomposition is possible, theoretically, its representation in the time domain becomes quickly usable, as shown in Figure 1.



Figure 1: Vibration signal decomposition into two harmonic components

To represent the vibration signal in a usable form, we tried to represent in a diagram of amplitudefrequency spectrum [26, 30]. Like any movement, vibration is characterized by relations between the three variables, involving frequency.

$$x = \frac{v}{2\pi f} = \frac{\gamma}{(2\pi f)^2}$$
(4)

$$x = \frac{\gamma}{2\pi f} = 2\pi f . x \tag{5}$$

$$\gamma = 2\pi f_V = (2\pi f)^2 x \tag{6}$$

It is apparent from equations (4), (5) and (6) the following relations among the vibration modules:

$$|X| = \frac{|V|}{\omega} = \frac{|A|}{\omega^2} \tag{7}$$

$$V|=|X|\cdot\omega=\frac{|A|}{\omega} \tag{8}$$

$$A| = |V| \cdot \omega = |X| \cdot \overset{2}{\omega}$$
(9)

Equations (7), (8) and (9) show the importance of the choice of the physical quantity to be measured by monitoring a rotating machine, as is the case of a gas turbine have generally consisted a rotor, and bonds of a structure, as shown in Figure 2.



Figure 2: Elements of a rotating machine

Any abnormality affecting a rotating machine (unbalance, imbalance phenomenon oil swirls, tree deformation, release bearing, bearing fault, electromagnetic anomaly in the stator or rotor of a motor, defective mesh, ...), translates to vibrations signals. Whose frequencies correspond to the occurrence of forces that induce and their harmonic frequencies (multiples of frequencies of occurrence). The overall measure used to quantify the default spectral analysis allows qualifying. The forces applied to the system are the inertia forces; they operate at the center of gravity.

$$\vec{F} = m \cdot \frac{d^2 \overline{OG}}{dt^2} \tag{10}$$

With , it is derived by: $\overrightarrow{OG} = \begin{cases} X = x + e\cos(\omega t) \\ Y = y + e\cos(\omega t) \end{cases}$

$$\frac{d^2 \overrightarrow{OG}}{dt^2} = \begin{cases} X'' = x'' - e\,\omega^2 \cos(\omega t) \\ Y'' = y'' - e\,\omega^2 \sin(\omega t) \end{cases}$$
(11)

$$\vec{F}_{i} = \begin{cases} mx'' - me\,\omega^{2}\cos(\omega t) \\ my'' - me\,\omega^{2}\sin(\omega t) \end{cases}$$
(12)

Stiffness forces it depends on Young's modulus, length of the shaft, the inertia of the shaft for conservative case, given by:

$$\vec{F} = -K\vec{OA} \Rightarrow \vec{OA} = \begin{cases} x \\ y \end{cases} \Rightarrow \vec{F}_{k} = \begin{cases} -kx \\ -ky \end{cases}$$
(13)

The damping forces are generally weak, the external depreciation is taken into consideration, it applies to the speed:

$$\overrightarrow{F_c} = -c.\frac{dOA}{dt}$$
(14)

Where .

The application of the principle of dynamics gives the equation of motion gives us:

 $\frac{d \overrightarrow{OA}}{dt} = \begin{cases} x' \\ y' \Rightarrow \overrightarrow{F_c} = \begin{cases} -cx' \\ -cy' \end{cases}$

$$\begin{cases} m\ddot{x} + c\dot{x} + kx = me\omega^2 \cos(\omega t) \\ m\ddot{y} + c\dot{y} + ky = me\omega^2 \sin(\omega t) \end{cases}$$
(15)

The spectrum can be obtained by applying the Fourier transform, which has the property of decomposing a signal into its complex basic components defined by their amplitude and frequency. This allows passing from one time representation to a spectral representation [22, 26]. The signal obtained by a vibration sensor in the time domain, by definition:

$$F(f) = \int_{-\infty}^{+\infty} f(t) \exp(-j2\pi f t) dt$$
 (16)

Where F(f) is the Fourier transform (FT)

Typically, the function f(t) is representative of a non-defined mathematical function by a simple signal, it should be sampled into discrete dots, and thereafter its spectrum can be calculated by substituting the Fourier integral of the algorithm by the Fast Fourier Transform (FFT), we use the following matches:

$$F(k\Delta f) = \frac{1}{N} \sum_{n=0}^{n=N-1} X(nt_e) \cdot \exp(-j2\pi \frac{kn}{N})$$
(17)

$$t \to nt_e f \to m\Delta f \quad dt \to t_e \qquad \int_{-\infty}^{\infty} \to \sum_{-\infty}^{\infty}$$

$$F(n-k) = \frac{1}{N} \sum_{n=0}^{n=N-1} X(nt_e) \cdot \exp(-j.2\pi \cdot (N-k) \cdot \frac{n}{N})$$

= F(-k) exp(-j.2\pi \cdot n) = F(-k)

With $\exp(j.2\pi n) = 1$ and |F(-k)| = |F(k)|.

So we have:

$$F(N-k) = |F(-k)| \tag{18}$$

Where t_e is the time sample signal, n is the number of the sample $0 \rightarrow N$, k is the number of the frequency line, Δf is the interval between two frequency lines (the time sample $\theta = N t_e$),

 $f_e = \frac{1}{t_e}$ is the sampling frequency of the signal and $f_e = 2.f_{max}$ is the maximum frequency of analysis. Generally, the decibel level L of vibration is giving by:

$$Lv = 10 \log_{10} \frac{V}{V_{ref}}$$
 (19)

3. Industrial application

In this work, we have examined a gas turbine installed in the DMLL management service SONATRACH, LAGHOUAT, Algeria. Measurements were performed on the gas turbine GE MS 3002, from our mobile accelerometers, as shown in Figure 3, for both levels (1 and 4) install us three sensors accelerometers positions (Horizontal, Vertical and Axial) at each level.



Figure 3: Sensors position installed for gas turbine monitoring

After several tests conducted on the site, the obtained spectra shown in Figures 4, 5, 6, 7, 8 and 9, its analysis shows that the energy of the fundamental signal which implies an effect of the "unbalance" and in that the axial first and second angular misalignment which manifest. An overview of GDE (Gear Defect Evaluation) values calculated from the measured bearing number 1, shows a GDA relatively high (about 0.54). This is synonymous with a problem of guiding the shaft in its bearing. It is noted that the phase difference between the vertical and horizontal position is 51.14° in the case of unbalance of phase around 90°, because the error of our sensor location with translational 39.86° from the horizontal, this operation due to the congestion prevents the base level, so the spectrum manifest in the effect of an unbalance. The recommendations proposed by this work, are to check the alignment between the turbine and gearbox and after checking the alignment balancing on site will be programmed to reduce vibration.



Figure 4: Signal tests RMS = 583.6 mg / Peak to Peak = 3849.2 mg / Crest Factor = 3.7 / kurtosis = 2.9 / Velocity = 5.3 mm



Figure 6: Spectrum obtained with FFT 0- 10 000 Hz and Acceleration RMS: 477.6 mg



Figure 8: Spectrum obtained with FFT 0-10 000 Hz and acceleration RMS: 222.6 mg



Figure 5: Spectrum obtained with FFT 0-10 000 Hz and acceleration RMS: 580.1 mg



Figure 7: Spectrum obtained with FFT 0-10 000 Hz and acceleration RMS: 222.6 mg



Figure 9: Spectrum obtained with FFT 0-10 000 Hz and acceleration RMS: 223.3 mg

A default of unbalance is indicated by a high amplitude component in the rotational frequency of the rotor in the radial direction, axially sometimes in the case of rotor cantilever. Also, an amplitude which can vary greatly with the speed of rotation with a phase shift close to 90 $^{\circ}$ between the two components corresponding to orthogonal radial measurement points on the same bearing of the rotor.

The unbalance is called "static" or "dynamic" according to the levels of order 1 of the rotation frequency will vibrate for a given phase or in phase opposition radial direction. A speed: 7100 r / min speed vibration reach Vib value = 10.8mm / s and a phase shift $\varphi = 59.49$. We perform a verification

with a launch mass fixation chosen (M = 170g) and start the machine. A speed V = 7113 rev / min, vibration velocity V = 4.48 mm / s phase $\varphi = 20.78^\circ$, as shown in Figures 10 and 11.



Figure 10: Vibration amplitude depending on the speed

Figure 11: Vibration phase depending on the speed

Vibration is decreased to the value 4.48 mm / s, and can not decrease to a value less than the obtained value (4.48mm / s) because the fixing screws do not support the larger than the selected mass weight (M = 170g). The proposed recommendations are the vibration monitoring of gas turbine which is observed during my internship some shortcomings in the present trend of defects after the intervention of the maintenance group (cell vibration) despite its expert performance management vibration analysis and processes.

4. Conclusion

The objective of this work is to provide the elements needed to monitor the vibration behavior in gas turbine and to defining a diagnosis approach to this examined system. The defects detection, such as flaws unbalance and misalignment and other alters the structure of the signals, the amplitude can increase and modulation amplitude and phase occur, the signals collected were found in different areas of analysis, spectral domain, domain transfer function analysis and the factors. The vibrations analysis proposed in this paper, based on defect signal evaluation, contains all information concerning the state of the mechanical parts of the studied gas turbine. The difficulty lies in the analysis of vibration signals and the identification of related components to be monitored. Our investigation in this practice study, with the presence of unbalance defects (major problem in rotating machines), consist to detect and locate such defects using spectral analysis in a low frequency range. The proposed tests based on defect signal evaluation gave good and efficient results.

5. References

- [1] Barella S., Boniardi M., Cincera S., Pellin P., Degive X., Gijbels S., Failure analysis of a third stage gas turbine blade. Engineering Failure Analysis, Volume 18, Issue 1, January 2011, Pages 386-393.
- [2] Chen Y.C., Chyanbin Hwu, Boundary element method for vibration analysis of two-dimensional anisotropic elastic solids containing holes, cracks or interfaces. Engineering Analysis with Boundary Elements, Volume 40, March 2014, Pages 22-35.

- [3] Enis Cetin A., Tom Pearson C., Akin Sevimli R., System for removing shell pieces from hazelnut kernels using impact vibration analysis. Computers and Electronics in Agriculture, Volume 101, February 2014, Pages 11-16.
- [4] Farrahi G.H., Tirehdast M., Masoumi Khalil Abad E., Parsa S., Motakefpoor M., Failure analysis of a gas turbine compressor. Engineering Failure Analysis, Volume 18, Issue 1, January 2011, Pages 474-484.
- [5] František Trebuňa, František Šimčák, Jozef Bocko, Róbert Huňady, Miroslav Pástor, Complex approach to the vibrodiagnostic analysis of excessive vibration of the exhaust fan. Engineering Failure Analysis, Volume 37, February 2014, Pages 86-95.
- [6] Hafaifa Ahmed, Mouloud Guemana, and Attia Daoudi, Vibration supervision in gas turbine based on parity space approach to increasing efficiency. Journal of Vibration and Control, doi: 10.1177/1077546313499927.
- [7] Hafaifa Ahmed, Mouloud Guemana and Attia Daoudi, Fault detection and isolation in industrial systems based on spectral analysis diagnosis. Intelligent Control and Automation Journal, 2013, February 2013, Vol. 4, pp. 36-41.
- [8] Hafaifa Ahmed, Ahmed Zohair Djeddi and Attia Daoudi, Fault detection and isolation in industrial control valve based on artificial neural networks diagnosis. Journal of Control Engineering and Applied Informatics CEAI, vol.15, no.3 pp. 61-69, 2013.
- [9] Hafaifa Ahmed, Attia Daoudi and Kouider Laroussi, Application of fuzzy fault detection and isolation approach to the compression system surge. ACTA Press, Control and Intelligent Systems. July 2011, vol. 39, No. 3, pp. 151-158.

[10] Hafaifa Ahmed, Attia Daoudi and Mouloud Guemana, SCADA for Surge Control: Using a SCADA Network to Handle Surge Control in Gas Suppression Systems in Pipelines. Control Global journal ISA Transactions, March 2011, vol. 24 no. 3, pp.69-71, 2011.

- [11] Hafaifa Ahmed, Ferhat Laaouad and Kouider Laroussi, A Numerical Structural Approach to Surge Detection and Isolation in Compression Systems using Fuzzy Logic Controller. International Journal of Control, Automation, and Systems, (IJCAS), February 2011, vol. 09, no. 01, pp. 69-79.
- [12] Hafaifa Ahmed, Kouider Laroussi and Ferhat Laaouad, Robust fuzzy fault detection and isolation approach applied to the surge in centrifugal compressor modeling and control. Fuzzy Information and Engineering, March 2010, vol.2 no.1, pp.49-73. URL:

[13] Hafaifa Ahmed, Ferhat Laaouad and Kouider Laroussi, Fuzzy logic approach applied to the surge detection and isolation in centrifugal compressor. Automatic Control and Computer Sciences, 2010, vol. 44, no. 1, pp. 53–59.

- [14] Hafaifa Ahmed, Ferhat Laaouad and Kouider Laroussi, Fuzzy modelling and control for detection and isolation of surge in industrial centrifugal compressors. Automatic Control Journal of the University of Belgrade 2009, vol.19, no.1, pp.19-26.
- [15] Halimi D, Hafaifa A, Bouali E. Maintenance actions planning in industrial centrifugal compressor based on failure analysis. Eksploatacja i Niezawodnosc – Maintenance and Reliability, January 2014; vol. 16, no. 1, pp. 17–21.
- [16] Jianfu Hou, Bryon J Wicks, Ross A Antoniou, An investigation of fatigue failures of turbine blades in a gas turbine engine by mechanical analysis. Engineering Failure Analysis, Volume 9, Issue 2, April 2002, Pages 201-211.
- [17] Jung-Hun Park, Hyun-Yong Park, Seok-Yong Jeong, Sang-Il Lee, Young-Ho Shin, Jong-Po Park, Linear vibration analysis of rotating wind-turbine blade. Current Applied Physics, Volume 10, Issue 2, Supplement, March 2010, Pages S332-S334.
- [18] Liu W.Y., The vibration analysis of wind turbine blade-cabin-tower coupling system. Engineering Structures, Volume 56, November 2013, Pages 954-957.

- [19] Martins R.F., Branco C.M., Gonçalves-Coelho A.M., Edgar C. Gomes, A failure analysis of exhaust systems for naval gas turbines. Part II: Design changes. Engineering Failure Analysis, Volume 16, Issue 4, June 2009, Pages 1324-1338.
- [20] Majcher K. and Wójcicki Z., Kinematically excited parametric vibration of a tall building model with a TMD—Part 1: Numerical analyses. Archives of Civil and Mechanical Engineering, Volume 14, Issue 1, January 2014, Pages 204-217.
- [21] Mouloud Guemana, Ahmed Hafaifa and Slimane Aissani, New approach of calibration in orifice flow meters applied in industrials gas pipeline systems to savings on maintenance costs, Hydrocarbon Processing Journal. August 2011, vol. 90, no. 8, pp. 63-68.
- [22] Mouloud Guemana, Slimane Aissani and Ahmed Hafaifa, Flow Measurement and Control in Gas Pipeline System using Intelligent Sonic Nozzle Sensor. Studies in Informatics and Control (SIC), June 2011, vol. 20, no. 02, pp. 85-96.
- [23] Missoum Lakhdar, Djermane Mohammed, Labbaci Boudjemâa, Abdeldjebar Rabiâ, Moudden Bachir, Damages Detection in a Composite Structure by Vibration Analysis. Energy Procedia, Volume 36, 2013, Pages 888-897.
- [24] Both Process and Vibration Data. Proceedings of the 3rd Gas Processing Symposium, 2012, Pages 334-343.Othman M., Najjar Y.S.H., Abou-Arab T.W., Fuel effect on induced vibration in gas turbine engines. Fuel, Volume 67, Issue 3, March 1988, Pages 321-326.
- [25] Osama Ashour, Abdurrahman Khalidi, Ever Fadlun, Nicola Giannini, Marco Pieri, Alberto Ceccherini, Online Monitoring of Gas Turbines to Improve Their Availability, Reliability, and Performance Using Pham-Ngoc Thach, Han-Long Liu, Gang-Qiang Kong, Vibration analysis of pile-supported embankments under high-speed train passage. Soil Dynamics and Earthquake Engineering, Volume 55, December 2013, Pages 92-99.
- [26] Poursaeidi E., Babaei A., Behrouzshad F., Mohammadi Arhani M.R., Failure analysis of an axial compressor first row rotating blades. Engineering Failure Analysis, Volume 28, March 2013, Pages 25-33.
- [27] Qu S., Fu C.M., Dong C., Tian J.F., Zhang Z.F., Failure analysis of the 1st stage blades in gas turbine engine. Engineering Failure Analysis, Volume 32, September 2013, Pages 292-303.
- [28] Robert T. Johnston, Real-time analysis system for gas turbine ground test acoustic measurements. ISA Transactions, Volume 42, Issue 4, October 2003, Pages 513-523.

[29] Saeed R.A., Galybin A.N., Popov V., 3D fluid–structure modelling and vibration analysis for fault diagnosis of Francis turbine using multiple ANN and multiple ANFIS. Mechanical Systems and Signal Processing, Volume 34, Issues 1–2, January 2013, Pages 259-276.

- [30] Sandeep Kumar, Niranjan Roy, Ranjan Ganguli, Monitoring low cycle fatigue damage in turbine blade using vibration characteristics. Mechanical Systems and Signal Processing, Volume 21, Issue 1, January 2007, Pages 480-501.
- [31] Xiang Xie, Guoyong Jin, Yuquan Yan, S.X. Shi, Zhigang Liu, Free vibration analysis of composite laminated cylindrical shells using the Haar wavelet method. Composite Structures, Volume 109, March 2014, Pages 169-177.